

Abstract

The invention relates to a rotor for a turbo engine, in particular for a gas turbine.

The rotor has a rotor base body (11) and a plurality of rotor blades (12) distributed around the circumference of the rotor base body (11).

According to this invention, the rotor base body (11) is formed by at least one ring-shaped element (13, 14) made of a metal matrix composite material, whereby the rotor blades (12) are attached to the rotor base body (11) via footing (18) so that the footing is positioned in a fiber-free area of the rotor base body.

(Fig. 2)